



Autonomous Vehicle Convoy Formation Control with Size/Shape Switching for Automated Highways

H. B. Jond^{*a}, J. Platoš^a, Z. Sadreddini^b

^a Department of Computer Science, VSB – Technical University of Ostrava, Ostrava, Czech Republic

^b Department of Computer Engineering, Istanbul Arel University, Istanbul, Turkey

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ABSTRACT

Today's semi-autonomous vehicles are gradually moving towards full autonomy. This transition requires developing effective control algorithms for handling complex autonomous tasks. Driving as a group of vehicles, referred to as a convoy, on automated highways is a highly important and challenging task that autonomous driving systems must deal with. This paper considers the control problem of a vehicle convoy modeled with linear dynamics. The convoy formation requirement is presented in terms of a quadratic performance index to minimize. The convoy formation control is formulated as a receding horizon linear-quadratic (LQ) optimal control problem. The receding horizon control law is innovatively defined via the solution to the algebraic Riccati equation. The solution matrix and therefore the receding horizon control law are obtained in the closed-form. A control architecture consisting of four algorithms is proposed to handle formation size/shape switching. The closed-form control law is at the core of these algorithms. Simulation results are provided to justify the models, solutions, and proposed algorithms.

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1. INTRODUCTION

Soon, public roads will host the massive deployment of autonomous vehicles. An autonomous convoy is a group of networked autonomous vehicles maintaining a formation (Figure 1). The formation control is defined as designing control inputs for the vehicles so that they form and maintain a pre-defined geometric shape. The line (or linear) formation, so-called platooning, concerns only longitudinal coordinated control of networked autonomous vehicles [1]. Convoy formation control requires both longitudinal and lateral coordination of vehicles [2]. Convoy control algorithms are essential for vehicle maneuvers such as lane change and overtaking on automated highways. Intelligent Traffic Management Systems [3] can also use these algorithms for city traffic.

Many algorithms from multi-agent systems are used in autonomous vehicle convoy formation control. Classical approaches which include leader-follower [4], virtual-structure [5] and behavior-based [6, 7] comprise a

significant amount of research effort in multi-agent systems formation control.

While the leader-follower approach is a popular design for the formation control, there are limitations. The loss of the leader or the leader being perturbed by some disturbances causes the entire group formation to fail [8]. On the contrary, the formation control can be leaderless, where all agents have the same role within the team.

In this study, we intend to derive an optimality-based formation control strategy for a leaderless autonomous

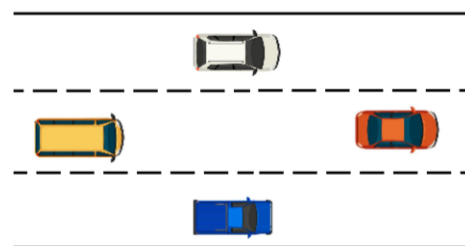


Figure 1. A convoy in a diamond formation

*Corresponding Author Institutional Email: [hossein.barghi.jond@vsb.cz](mailto:hosseini.barghi.jond@vsb.cz)
(H. B. Jond)

convoy. The optimality-based approach has received attention in robotic car literature (see, e.g., [9, 10]). Particularly the LQ modeling of formation control is highly interesting due to the analytic tractability of LQ problems. In this approach, the formation objective, which is to drive multiple agents to achieve a prescribed constraint on their states, is implemented by using a quadratic performance index through the use of graph theory [11]. The multi-agent system dynamics is modeled as a controllable linear system.

Depending on the available information structure, open-loop [12], state-feedback [13] and receding horizon [14] control structures can be investigated for formation control. The receding horizon or model predictive controller uses the open-loop control signals to implement an online algorithm that predicts the system's output based on current states and system models. It has become the most popular feedback strategy in industrial applications.

In the operation field, when obstacles or boundaries are detected along the formation's path, the formation is able to squeeze through obstacles by switching to suited patterns that are chosen among a collection of formations. In some other cases, a safe formation control strategy scales the formation shape (by a size switching strategy). By adjusting the scale, the formation can grow and shrink as necessary to accommodate and avoid obstacles in its surrounding area. An illustrative example is given in [15], where a group of agents is to traverse a narrow passage while maintaining a desired triangular shape.

In this paper, the convoy formation control is addressed as a receding horizon LQ optimal control problem. Under this framework, the matrix Riccati equation must be solved. Obtaining a solution to this equation is not generally straightforward. This type of matrix equation has been discussed in detail in [16]. Innovatively, the receding horizon control law is constructed via the algebraic Riccati equation, which leads to obtaining the control law in the closed-form. Besides, a control scheme is developed to deal with the formation size/shape switching under the receding horizon LQ framework. This scheme consists of four control modes in which each control mode is run under an algorithm constructed via the closed-form control law.

The remainder of this paper is organized as follows. The system models and LQ optimal control formulation of the formation control problem are introduced in Section 2. The receding horizon control design is presented in Section 3. The control scheme and corresponding algorithms for formation size/shape switching are developed in Section 4. The simulation results are shown in Section 5. The conclusion is given in Section 6.

2. CONVOY FORMATION STATEMENT

Vehicle dynamics is highly nonlinear. Hierarchical control architectures at their top-level consider a simplified dynamics model of the system and shift the nonlinearity to their lower levels. An appropriate simple dynamical model for a vehicle is the double integrator model. It simplifies vehicle dynamics as a point mass governed by Newton's laws [17].

Consider a convoy of m networked vehicles, each of which is described by a double integrator dynamics. Let $\mathbf{q}_i, \mathbf{v}_i, \mathbf{u}_i \in \mathbb{R}^2$ be the coordinates, velocity and control input vectors for vehicle i ($i = 1, \dots, m$), respectively. Define $\mathbf{z} = [\mathbf{q}_1^T, \dots, \mathbf{q}_m^T, 1, \mathbf{v}_1^T, \dots, \mathbf{v}_m^T]^T$ and $\mathbf{u} = [\mathbf{u}_1^T, \dots, \mathbf{u}_m^T]^T$. Vectors $\mathbf{z} \in \mathbb{R}^{4m+1}$ and $\mathbf{u} \in \mathbb{R}^{2m}$ are system state and control input vectors, respectively. The system dynamics can be expressed as

$$\dot{\mathbf{z}} = \mathbf{A}\mathbf{z} + \mathbf{B}\mathbf{u} \quad (1)$$

where $\mathbf{A} = \begin{bmatrix} \mathbf{0} & \mathbf{I}_{2m} \\ \mathbf{0} & \mathbf{0} \end{bmatrix}$, $\mathbf{B} = [\mathbf{0}_{2m}, \mathbf{0}_{2m \times 1}, \mathbf{I}_{2m}]^T$, and \mathbf{I} is the identity matrix of appropriate dimension.

Networked systems such as a convoy of vehicles exchange information via a communication network. The information flow over the communication network can be modeled with graph theory. A directed graph $\mathcal{G} = (\mathcal{V}, \mathcal{E})$ consists of a set of vertices $\mathcal{V} = \{1, 2, \dots, m\}$ and a set of edges $\mathcal{E} \subseteq \{(i, j) : i, j \in \mathcal{V}\}$ containing ordered pairs of distinct vertices. For the formation control, the set of vertices \mathcal{V} corresponds to the vehicles and then the set of edges \mathcal{E} represents the interconnections. Each edge $(i, j) \in \mathcal{E}$ is assigned with a weight $\mu_{ij} > 0$.

Assumption 1. Formation graph \mathcal{G} is connected, i.e., for every pair of vertices $i, j \in \mathcal{V}$, from i to j for all $j = 1, \dots, m, j \neq i$, there exists a path of (undirected) edges from \mathcal{E} .

The graph Laplacian $\mathbf{L} \in \mathbb{R}^m$ is defined as

$$\mathbf{L} = \mathbf{D}\mathbf{W}\mathbf{D}^T \quad (2)$$

where $\mathbf{D} \in \mathbb{R}^{m \times |\mathcal{E}|}$ is the incidence matrix and $\mathbf{W} = \text{diag}(\mu_{ij}) \in \mathbb{R}^{|\mathcal{E}|}$ is a diagonal weight matrix. \mathbf{D} 's uv th element is 1 if the node u is the head of the edge v , -1 if the node u is the tail, and 0, otherwise.

The Kronecker product \otimes can be used to extend the dimension. The 2-dimensional graph Laplacian $\mathcal{L} \in \mathbb{R}^{2m}$ is defined as

$$\mathcal{L} = \mathbf{L} \otimes \mathbf{I}_2 \quad (3)$$

Based on the properties of the Kronecker product, \mathcal{L} can be rearranged as

$$\mathcal{L} = \mathbf{D}\mathbf{W}\mathbf{D}^T \otimes \mathbf{I}_2 = (\mathbf{D} \otimes \mathbf{I}_2)(\mathbf{W} \otimes \mathbf{I}_2)(\mathbf{D} \otimes \mathbf{I}_2)^T = \mathcal{D}\mathcal{W}\mathcal{D}^T \quad (4)$$

where $\mathcal{D} = \mathbf{D} \otimes \mathbf{I}_2$ and $\mathcal{W} = \mathbf{W} \otimes \mathbf{I}_2$.

The graph Laplacian is symmetric, positive semidefinite and holds the sum-of-squares property [18]:

$$\mathbf{z}^T \mathbf{L} \mathbf{z} = \sum_{(i,j) \in \mathcal{E}} \mu_{ij} \|\mathbf{q}_i - \mathbf{q}_j\|^2 \quad (5)$$

where $\|\cdot\|$ is the Euclidean norm in \mathbb{R}^2 .

The formation requirement according to the information graph can be expressed as

$$\sum_{(i,j) \in \mathcal{E}} \mu_{ij} \left(\|\mathbf{q}_i - \mathbf{q}_j - \mathbf{d}_{ij}\|^2 + \|\mathbf{v}_i - \mathbf{v}_j\|^2 \right) \rightarrow 0 \quad (6)$$

where $\mathbf{d}_{ij} \in \mathbb{R}^n$ is the desired distance vector between two neighbor vehicles i and j . Using the property of sum-of-squares, (6) can be transformed into the following matrix form

$$\sum_{(i,j) \in \mathcal{E}} \omega_{ij} \left(\|\mathbf{q}_i - \mathbf{q}_j\|^2 - 2(\mathbf{q}_i - \mathbf{q}_j)^T \mathbf{d}_{ij} + \|\mathbf{d}_{ij}\|^2 + \|\mathbf{v}_i - \mathbf{v}_j\|^2 \right) = \mathbf{q}^T \mathbf{L} \mathbf{q} - 2\mathbf{q}^T \mathcal{D} \mathbf{W} \mathbf{d} + \mathbf{d}^T \mathbf{W} \mathbf{d} + \mathbf{v}^T \mathcal{L} \mathbf{v} = \mathbf{z}^T \mathbf{Q} \mathbf{z} \quad (7)$$

where $\mathbf{Q} = \begin{bmatrix} \mathcal{L} & -\mathcal{D} \mathbf{W} \mathbf{d} & \mathbf{0} \\ -(\mathcal{D} \mathbf{W} \mathbf{d})^T & \mathbf{d}^T \mathbf{W} \mathbf{d} & \mathbf{0} \\ \mathbf{0} & \mathbf{0} & \mathcal{L} \end{bmatrix}$, $\mathbf{d} = \text{col}(\mathbf{d}_{ij})$ and $\text{col}(\cdot)$ stands for "column vector". As $\mathbf{z}^T \mathbf{Q} \mathbf{z} \geq 0$, matrix \mathbf{Q} is positive semidefinite.

A performance index for the convoy formation control is defined as

$$J = \mathbf{z}(t_f)^T \mathbf{Q}_f \mathbf{z}(t_f) + \int_0^{t_f} (\mathbf{z}^T \mathbf{Q} \mathbf{z} + \mathbf{u}^T \mathbf{R} \mathbf{u}) dt \quad (8)$$

where $\mathbf{Q}_f = \begin{bmatrix} \mathcal{L}_f & -\mathcal{D} \mathbf{W}_f \mathbf{d} & \mathbf{0} \\ -(\mathcal{D} \mathbf{W}_f \mathbf{d})^T & \mathbf{d}^T \mathbf{W}_f \mathbf{d} & \mathbf{0} \\ \mathbf{0} & \mathbf{0} & \mathcal{L}_f \end{bmatrix}$, $\mathcal{L}_f = \mathcal{D} \mathbf{W}_f \mathcal{D}^T$, $\mathbf{W}_f = \mathbf{W}_f \otimes \mathbf{I}_n$, $\mathbf{W}_f = \text{diag}(\omega_{ij}) \in \mathbb{R}^{|\mathcal{E}|}$, $\omega_{ij} > 0$, t_f is the fixed finite horizon length and $\mathbf{R} \in \mathbb{R}^{2m}$ is diagonal positive definite ($\mathbf{R} > \mathbf{0}$). Matrices \mathbf{W} and \mathbf{R} represent the penalties on the formation-velocity errors and control effort during the entire formation control process, respectively. Matrix \mathbf{W}_f represents the penalty on the terminal formation-velocity errors.

The convoy formation control objective is to design the control input vector \mathbf{u} to minimize the performance index J for the underlying system dynamics (1). The formation control problem under the framework of LQ optimal control formulation converts to the symmetric Riccati differential equation problem, which is stated in the following theorem. The proof can be found in [19].

Theorem 1. For the formation control defined as the LQ optimal control problem (1) and (8), the open-loop solution is given by

$$\mathbf{u} = -\mathbf{R}^{-1} \mathbf{B}^T \mathbf{P} \mathbf{z} \quad (9)$$

where \mathbf{P} is the solution to the Riccati differential equation;

$$\dot{\mathbf{P}} + \mathbf{P} \mathbf{A} + \mathbf{A}^T \mathbf{P} - \mathbf{P} \mathbf{S} \mathbf{P} + \mathbf{Q} = \mathbf{0}, \quad \mathbf{P}(t_f) = \mathbf{Q}_f \quad (10)$$

and $\mathbf{S} = \mathbf{B} \mathbf{R}^{-1} \mathbf{B}^T$.

Matrix \mathbf{P} is symmetric positive semidefinite. In general, (10) must be solved numerically by using the terminal value and backward iteration.

3. RECEDING HORIZON CONTROL

Fixed horizon control problems suffer from the main drawback that unexpected changes in the system that may happen at a future time cannot be included in the model. This issue is addressed by the idea of receding horizon control [20]. Let $0 < \delta < t_f$ denote the sampling period. In the receding horizon control, the current control law \mathbf{u} is obtained by solving the open-loop optimal control problem (1) and (8) at each sampling instant t for the interval $[t, t + t_f]$. Then, \mathbf{u} and the corresponding system trajectory \mathbf{z} is used until the next sampling time $t + \delta$ arrives. In this method, at each time instant t the current state vector \mathbf{z} is considered as the initial state. The open-loop control signal \mathbf{u} minimizes the following receding horizon performance index

$$j = \mathbf{z}(t + t_f)^T \mathbf{Q}_f \mathbf{z}(t + t_f) + \int_t^{t+t_f} (\mathbf{z}^T \mathbf{Q} \mathbf{z} + \mathbf{u}^T \mathbf{R} \mathbf{u}) dt \quad (11)$$

Following [14], the receding horizon control for the formation control problem is defined as

$$\bar{\mathbf{u}} = -\mathbf{R}^{-1} \mathbf{B}^T \mathbf{P}(0) \mathbf{z} \quad (12)$$

The closed-loop system is

$$\dot{\mathbf{z}} = \mathbf{A}_{cl}(0) \mathbf{z} \quad (13)$$

where $\mathbf{A}_{cl}(0) = \mathbf{A} - \mathbf{S} \mathbf{P}(0)$ is the closed-loop system matrix.

As it is seen from (12) calculation of the signal $\bar{\mathbf{u}}$ needs the solution matrix $\mathbf{P}(0)$ of the Riccati differential Equation (10). In a particular case in which $t_f \rightarrow \infty$, \mathbf{P} approaches a finite constant \mathcal{P} where it satisfies the algebraic Riccati equation (ARE)

$$\mathcal{P} \mathbf{A} + \mathbf{A}^T \mathcal{P} - \mathcal{P} \mathbf{S} \mathcal{P} + \mathbf{Q} = \mathbf{0} \quad (14)$$

In the system dynamics (1) if (\mathbf{A}, \mathbf{B}) is stabilizable then \mathcal{P} is a stabilizing solution to (10). Moreover, \mathbf{P} approaches to \mathcal{P} as $t \rightarrow -\infty$. Therefore, in this paper, we propose the idea that solution \mathcal{P} to the ARE (14) can be used in (12) instead of $\mathbf{P}(0)$. Consequently, the receding horizon control for the formation control problem in (1) and (11) is redefined as

$$\bar{\mathbf{u}} = -\mathbf{R}^{-1} \mathbf{B}^T \mathcal{P} \mathbf{z} \quad (15)$$

In the next theorem, the closed-form solution of the algebraic Riccati Equation (14) is presented. Before, the following definition is introduced.

Definition 1. Let the real symmetric matrix \mathbf{M} be positive semidefinite. The square root of \mathbf{M} is denoted by $\mathbf{M}^{\frac{1}{2}}$ and

satisfies $\mathbf{M} = \mathbf{M}^{\frac{1}{2}} \mathbf{M}^{\frac{1}{2}}$. Any symmetric, positive semidefinite matrix has a unique symmetric, semidefinite square root [21].

Theorem 2. The unique symmetric solution to the algebraic Riccati Equation (14) is in the following form:

$$\mathcal{P} = \begin{bmatrix} \mathbf{MR}^{-1}\mathbf{N} & -\mathbf{MN}^{-1}\mathcal{D}\mathcal{W}\mathbf{d} & \mathbf{N} \\ * & * & * \\ \mathbf{N} & -\mathbf{RN}^{-1}\mathcal{D}\mathcal{W}\mathbf{d} & \mathbf{M} \end{bmatrix} \quad (16)$$

where $\mathbf{M} = (2\mathbf{NR} + \mathbf{N}^2)^{\frac{1}{2}}$, $\mathbf{N} = (\mathcal{L}\mathbf{R})^{\frac{1}{2}}$ and * are the elements/blocks to be not concerned.

Proof. By ignoring the (artificially made) middle row of \mathbf{A} we see that (\mathbf{A}, \mathbf{B}) is stabilizable. As \mathbf{R} is positive definite, and \mathbf{Q} is symmetric positive semidefinite, there exists a unique symmetric solution to (14).

Assume that \mathcal{P} has the following form:

$$\mathcal{P} = \begin{bmatrix} \mathcal{P}_{11} & \mathcal{P}_{12} & \mathcal{P}_{13} \\ \mathcal{P}_{12}^T & \mathcal{P}_{22} & \mathcal{P}_{23} \\ \mathcal{P}_{13} & \mathcal{P}_{23}^T & \mathcal{P}_{33} \end{bmatrix} \quad (17)$$

Substituting \mathbf{A} , \mathbf{S} and \mathbf{Q} into (14) yields

$$\mathcal{P}_{13}\mathbf{R}^{-1}\mathcal{P}_{13} - \mathcal{L} = \mathbf{0} \quad (18)$$

$$2\mathcal{P}_{13} - \mathcal{P}_{33}\mathbf{R}^{-1}\mathcal{P}_{33} + \mathcal{L} = \mathbf{0} \quad (19)$$

$$\mathcal{P}_{11} - \mathcal{P}_{33}\mathbf{R}^{-1}\mathcal{P}_{13} = \mathbf{0} \quad (20)$$

$$\mathcal{P}_{23}\mathbf{R}^{-1}\mathcal{P}_{13} + (\mathcal{D}\mathcal{W}\mathbf{d})^T = \mathbf{0} \quad (21)$$

$$\mathcal{P}_{12} - \mathcal{P}_{33}\mathbf{R}^{-1}\mathcal{P}_{23}^T = \mathbf{0} \quad (22)$$

By solving these equations using the matrix square root definition we obtain (16). Finally, from (15) it can be seen that the calculation of $\bar{\mathbf{u}}$ is not dependent on the middle row elements/blocks of \mathcal{P} .

The closed-loop system matrix is redefined as $\bar{\mathbf{A}}_{cl} = \mathbf{A} - \mathbf{S}\mathcal{P}$. The system is asymptotically stable on the condition that all the eigenvalues of $\bar{\mathbf{A}}_{cl}$ have negative real parts. As $\bar{\mathbf{A}}_{cl}$ is a function of penalty matrices \mathbf{W} and \mathbf{R} , the asymptotic stability can be achieved by selecting appropriate \mathbf{W} and \mathbf{R} matrices.

Together with satisfying the asymptotic stability condition, matrices \mathbf{W} and \mathbf{R} should reflect the real-world requirements such as vehicle's energy consumption management. For example, if the vehicles have a sufficient amount of fuel in their tank or if they are near a gas station, a large \mathbf{W} and a small \mathbf{R} can be selected to emphasize the formation-velocity errors rather than the control effort. On the contrary, a small \mathbf{W} and a large \mathbf{R} can be selected for situations in which the vehicles are on an energy-saving policy. The best trade-off between the system performance (e.g. convoy formation) and control effort (e.g. energy consumption) should be taken.

4. SIZE/SHAPE SWITCHING

The receding horizon based controller calculates controls that drive each vehicle to acquire the desired relative position with respect to its neighbor(s) in the graph topology. During the process of acquiring a desired formation shape, at every time instant, there could be a change in the environment/system that might require a new desired formation shape to be acquired. For example, due to newly detected obstacles or failed vehicle(s) in the group, the initial formation topology would be subject to change in size/shape.

To handle size/shape switching a control architecture is presented in Figure 2. Based on the signal received from an external observer or a decision-maker vehicle equipped with appropriate sensors, a control mode is selected at each time instant. The control loop terminates when the formation control expectations are fulfilled.

Initially, the vehicle convoy is supposed to acquire the desired formation under the receding horizon control scheme. The receding horizon approach to the formation control problem is implemented using Algorithm 1. Here, the control commands are generated by the closed-form law in (15).

Algorithm 1. Receding Horizon Control

At each time instant t :

- 1: **Measure** the current state vector \mathbf{z}
 - 2: **Calculate** \mathbf{u} and \mathbf{z} for $[t, t + t_f]$
 - 3: **Apply** $\mathbf{u} = \bar{\mathbf{u}}$ for the period $[t, t + \delta]$
 - 4: **Update** $t \leftarrow t + \delta$
 - 5: **Repeat** the steps 1 to 4
-

Size switching is subject to change only at desired distances among neighboring vehicles. Within this control platform when a size switching signal is received the formation desired distances vector $\mathbf{d} = \text{col}(\mathbf{d}_{ij})$ is multiplied by a scalar α where α is the formation shape scale factor. In this case, recalculation of the open-loop

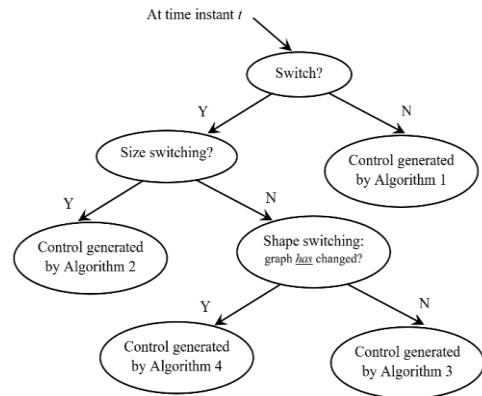


Figure 2. Proposed control structure

control \mathbf{u} for $[t, t + t_f]$ needs fewer computation resources. This is because the control signal can be rewritten as

$$\mathbf{u} = -\mathbf{R}^{-1}(\mathbf{N}\mathbf{q} - \mathbf{R}(\mathbf{N}^{-1})^T \mathbf{D}\mathbf{W}\mathbf{d} + \mathbf{M}\mathbf{v}) \quad (23)$$

and as it is seen only the middle term is subject to being multiplied by scalar α . The proposed size switching procedure is presented in Algorithm 2.

Algorithm 2. Size Switching

At each time instant t :

- 1: **Measure** the current state vector $\mathbf{z}(t)$
 - 2: **Update** \mathbf{d} as $\alpha\mathbf{d}$
 - 3: **Calculate** \mathbf{u} and \mathbf{z} for $[t, t + t_f]$
 - 4: **Apply** $\mathbf{u} = \bar{\mathbf{u}}$ for the period $[t, t + \delta]$
 - 5: **Update** $t \leftarrow t + \delta$
 - 6: **Repeat** the steps 1 to 5
-

In shape switching the vehicles might keep their neighbors but need to acquire different desired distances. In other words, formation graph \mathcal{G} does *not* change. In this case, similar to size switching only the middle term of the control signal is subject to the minor change of updating \mathbf{d} . The proposed shape switching procedure is presented in Algorithm 3.

Algorithm 3. Shape Switching (\mathcal{G} has *not* changed)

At each time instant t :

- 1: **Measure** the current state vector $\mathbf{z}(t)$
 - 2: **Update** \mathbf{d}
 - 3: **Calculate** \mathbf{u} and \mathbf{z} for $[t, t + t_f]$
 - 4: **Apply** $\mathbf{u} = \bar{\mathbf{u}}$ for the period $[t, t + \delta]$
 - 5: **Update** $t \leftarrow t + \delta$
 - 6: **Repeat** the steps 1 to 5
-

Shape switching might be subject to change on interconnections. Some vehicles might lose some of their immediate neighbors in the graph topology and some might acquire new neighbors (i.e., \mathcal{G} changes). Here, the \mathbf{D} , \mathbf{W} and \mathbf{d} parameters of the control signal must be updated. The proposed shape switching procedure when \mathcal{G} changes, is presented in Algorithm 4.

Algorithm 4. Shape Switching (\mathcal{G} has changed)

At each time instant t :

- 1: **Measure** the current state vector $\mathbf{z}(t)$
 - 2: **Update** \mathbf{D} , \mathbf{W} , \mathbf{d}
 - 3: **Calculate** \mathbf{u} and \mathbf{z} for $[t, t + t_f]$
 - 4: **Apply** $\mathbf{u} = \bar{\mathbf{u}}$ for the period $[t, t + \delta]$
 - 5: **Update** $t \leftarrow t + \delta$
 - 6: **Repeat** the steps 1 to 5
-

When designing the software interface of the control structure in Figure 2 (for example in Java), each different control mode can be addressed by using the abstract methods presented in Table 1.

TABLE 1. Abstract methods for different control procedures

Abstract method	Control generated by	Description
$control()$	Algorithm 1	Receding horizon control
$control(\alpha)$	Algorithm 2	Size switching, \mathbf{d} updates as $\alpha\mathbf{d}$
$control(\mathbf{d})$	Algorithm 3	Shape switching, \mathcal{G} has not changed, only \mathbf{d} updates
$control(\mathbf{D}, \mathbf{W}, \mathbf{d})$	Algorithm 4	Shape switching, \mathcal{G} has changed, \mathbf{D} , \mathbf{W} , \mathbf{d} update

5. SIMULATIONS

Simulations are carried out to analyze the proposed control architecture and algorithms. The experiments consist of five vehicles ($m = 5$). We consider a scenario in which at the time instants $t = 0s$ and $t = 7s$, the vehicles receive the signals $control()$ and $control(\mathbf{D}, \mathbf{W}, \mathbf{d})$, respectively. The corresponding information graphs are given in Figure 3, and their incidence matrices are, respectively:

$$\mathbf{D} = \begin{bmatrix} -1 & -1 & 0 & 0 \\ 1 & 0 & -1 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 1 & 0 & -1 \\ 0 & 0 & 0 & 1 \end{bmatrix}, \quad \mathbf{D} = \begin{bmatrix} 0 & -1 & 0 & 0 \\ 1 & 0 & -1 & 0 \\ 0 & 0 & 1 & 0 \\ -1 & 1 & 0 & -1 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

The weight matrices \mathbf{W} and \mathbf{R} are selected as the identity matrices. The desired offset vectors of the formation shape among the vehicles at $t = 0s$ are: $\mathbf{d}_{12} = \mathbf{d}_{23} = [-2, -4]^T$ and $\mathbf{d}_{14} = \mathbf{d}_{45} = [2, -4]^T$. These vectors mean that to form the desired shape; vehicle 1 and 2, as well as vehicle 2 and 3, have to achieve to relative distance -2 and -4 in the x and y position, respectively; vehicle 1 and 4, as well as vehicle 4 and 5, have to achieve to relative distance 2 and -4 in the x and y position, respectively. Moreover, all vehicles have to achieve zero relative velocities.

At time instant $t = 7s$, the communication link between vehicles 1 and 2 must break down and a new link between vehicles 2 and 4 must be built. The new desired offset vectors are: $\mathbf{d}_{24} = [2, 0]^T$, $\mathbf{d}_{14} = [2, -4]^T$, $\mathbf{d}_{23} = \mathbf{d}_{45} = [0, -4]^T$.

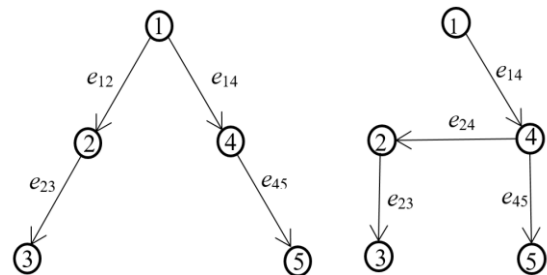


Figure 3. Information graph at $t = 0s$ (left) and $t = 7s$ (right)

The initial positions of the vehicles are set to: $\mathbf{q}_1 = [1,0]^T$, $\mathbf{q}_2 = [4,0]^T$, $\mathbf{q}_3 = [7,0]^T$, $\mathbf{q}_4 = [-1,0]^T$, $\mathbf{q}_5 = [-4,0]^T$. The initial velocities are set to: $\dot{\mathbf{q}}_1 = [0,2]^T$, $\dot{\mathbf{q}}_2 = [0,3]^T$, $\dot{\mathbf{q}}_3 = [0,1.5]^T$, $\dot{\mathbf{q}}_4 = [0,1]^T$, $\dot{\mathbf{q}}_5 = [0,2.5]^T$.

The finite horizon length is selected as $t_f = 7s$ and the sampling time for simulation/control is $0,1s$. The total running time of the control is $14s$. The vehicles' trajectories under signals $control()$ and $control(\mathbf{D}, \mathbf{W}, \mathbf{d})$ are shown in Figure 4. The vehicles are shown at $t = 0s$, $t = 5s$, and $t = 12s$. The results show that the desired formations among the vehicles are achieved under both signals at the end of their time interval. Time histories of the relative positions and velocities and also the control input of the individual vehicles are shown in Figure 5. It can be seen that all the formation control objectives are achieved.

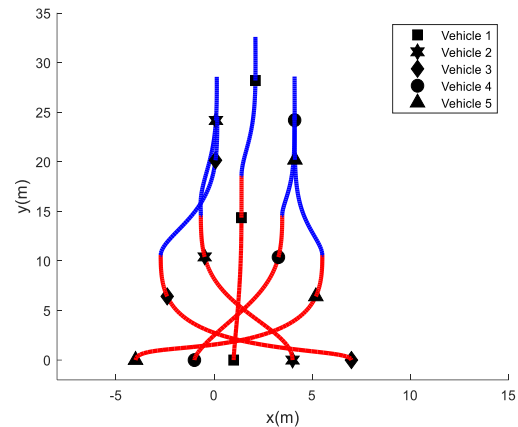


Figure 4. Vehicles' trajectories under $control()$ followed by $control(\mathbf{D}, \mathbf{W}, \mathbf{d})$ in red and blue, respectively

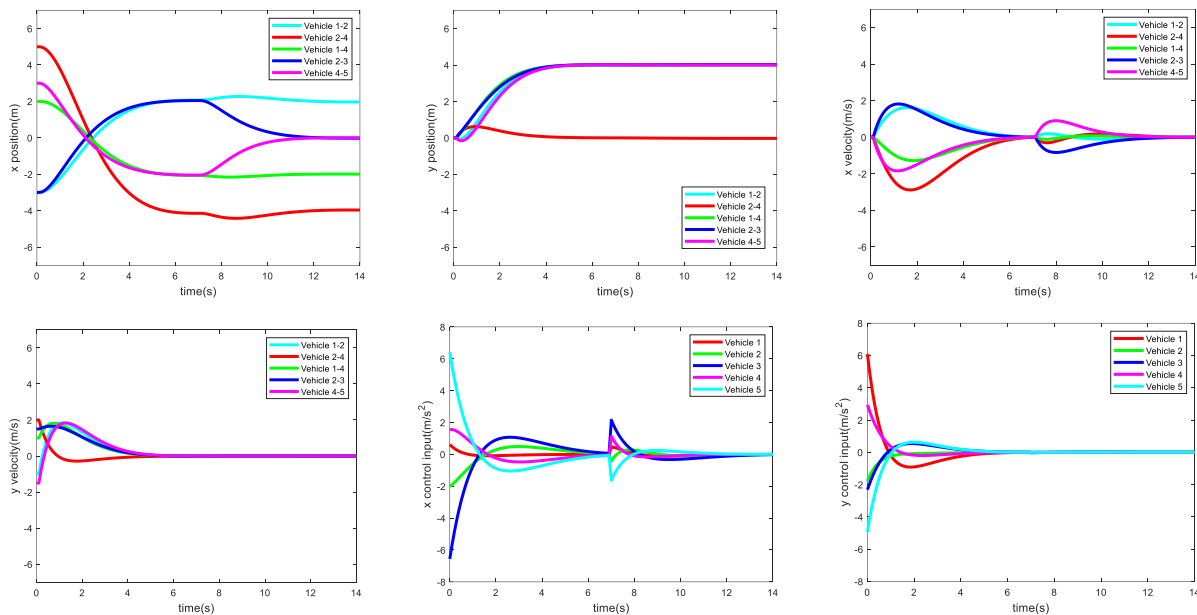


Figure 5. Time histories of the relative positions, velocities and control inputs under $control()$ followed by $control(\mathbf{D}, \mathbf{W}, \mathbf{d})$

6. CONCLUSION

In this paper, the convoy formation control problem was investigated under the receding horizon LQ optimal control framework. A closed-form solution was presented for the receding horizon LQ optimal formation control. The solution to the control law is a non-linear function of the graph Laplacian matrix and the formation desired distance vectors. A control architecture consisting of four control modes for formation size/shape switching problems was proposed. An algorithm based on the presented closed-form control law is developed for each control mode. The models, closed-form solution and proposed algorithms are approved by the simulations.

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Persian Abstract

چکیده

وسایل نقلیه نیمه خودکار و هوشمند امروزی به تدریج به سمت خودمختاری کامل حرکت می‌کنند. این انتقال مستلزم ایجاد الگوریتم‌های کنترل موثر برای انجام وظایف مختلط پیچیده است. رانندگی به عنوان گروهی از وسایل نقلیه، که به آنها کاروان گفته می‌شود، در بزرگراه‌های خودکار وظیفه‌ای بسیار مهم و چالش برانگیز است که سیستم‌های رانندگی مستقل باید از پس آن برآیند. در این مقاله مشکل کنترل کاروان وسیله نقلیه مدل شده با پویایی خطی در نظر گرفته شده است. نیاز تشکیل کاروان از نظر شاخص عملکرد درجه دوم ارائه می‌شود تا به حداقل برسد. کنترل تشکیل کاروان به عنوان یک مسئله کنترل بهینه خطی-درجه دوم LQ برای افق عقب‌سازی فرموله شده است. قانون کنترل افق در حال عقب‌نشینی از طریق راه‌حل معادله جبری ریکاتی تعریف شده است. ماتریس راه‌حل و بنابراین قانون کنترل افق در حال عقب‌نشینی به صورت بسته به دست می‌آید. یک معماری کنترلی متشکل از چهار الگوریتم برای کنترل تغییر شکل اندازه / شکل تشکیل شده است. قانون کنترل فرم بسته در هسته اصلی این الگوریتم‌ها است. نتایج شبیه‌سازی برای توجیه مدل‌ها، راه‌حل‌ها و الگوریتم‌های پیشنهادی ارائه شده است.